

Escape of rock-forming volatile elements and noble gases from planetary embryos

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Abstract

Large planetesimals and planetary embryos ranging from several hundred to a few thousand kilometers can develop magma oceans through mutual collisions, gravitational energy, and the heating of short-lived radioactive elements. During their solidification after the dissipation of the disk a steam atmosphere will be catastrophically outgassed and may be lost efficiently via hydrodynamic escape, as long as it does not condense. The escaping H-atoms that originate from the dissociation of H₂O and H₂ will drag heavier trace elements like noble gases such as Ne and Ar and outgassed moderately volatile rock-forming elements such as K, Na, Si, Mg, etc. into space. Under consideration of **various EUV flux evolution scenarios** of young solar-type stars, we apply an upper atmosphere hydrodynamic escape model that includes the dragging of heavier species by escaping H-atoms. We investigate the atmospheric/elemental escape and fractionation from planetary embryos with masses of $1 M_{\text{Moon}}$, $0.5 M_{\text{Mars}}$, $1 M_{\text{Mars}}$, and $1.5 M_{\text{Mars}}$ at different orbital distances between the orbits of Venus and Mars. Our results indicate that the steam atmospheres and the embedded trace elements will be lost efficiently before they condense for masses $\leq 0.5 M_{\text{Mars}}$ and orbital distances up to 1 AU. For heavier embryos of up to $1.5 M_{\text{Mars}}$ almost all of the considered steam atmospheres can be lost within ≈ 12 Myr, which lies within the time frame of the formation of the first Martian protocrust after ≈ 20 Myr,

i.e. for such steam atmospheres to be lost completely a shallow magma ocean must remain below the atmosphere, which might be achieved through frequent impacts onto the planetary embryo. The considered outgassed noble gases and rock-forming elements will be completely dragged away together with the steam atmospheres under the assumption that the trace elements will reach the thermosphere. For embryos with masses $\leq M_{\text{Moon}}$ the gravity is too weak for a dense atmosphere to build up for the high magma ocean related surface temperatures and all outgassed elements will escape immediately to space. For all considered planetary masses **and orbits** the loss rates of Ar and Ne are so high that there will be no fractionation of their isotopes. The studied planetary embryos, even though not isotopically fractionated, will therefore be severely depleted in noble gases and moderately volatile elements. **Hydrodynamic escape** might then also affect the final composition of terrestrial planets that accrete out of such planetary embryos, such as the volatile content and the Fe/Mg ratio of a planet.

Keywords: Atmospheres evolution, Solar radiation, Planetary formation

1. Introduction

During the early evolution of planetary systems large planetesimals and planetary embryos are formed out of dust and ice, while embedded in the circumstellar disks (Wood, 2005; Birnstiel et al., 2016; Blum, 2018). **Within the first 3 Myr following the birth of the solar system, ^{26}Al , the most abundant short-lived radioactive heat source capable of melting rocky material, becomes extinct.** Heating by short-lived radioisotopes such as ^{60}Fe and ^{26}Al determined the thermal history and interior structure of these planetary building blocks during the earliest stages of planetary formation (Lichtenberg et al., 2016a, 2018). The following thermo-mechanical evolution, like internal differentiation and rapid volatile outgassing, yields important implications for the final composition and hence the evolution of terrestrial planets. The accreting planetary bodies whose radii range from tens to hundreds of kilometers (Elkins-Tanton, 2012; Lichtenberg et al., 2016b, 2018) and more, experienced significant

and perhaps complete melting due to radiogenic heating from these short-lived radioisotopes (Urey, 1955; Fish et al., 1960; Lichtenberg et al., 2016b,a, 2018, 2019; Hin et al., 2017; Young et al., 2019) as well as during energetic accretionary impacts with other large planetesimals and planetary embryos (Safronov & Zvjagina, 1969; Wetherill, 1980; Tonks & Melosh, 1993).

The magma oceans on these planetary building blocks are responsible for differentiation and their resulting composition. During this hot magmatic phase thermodynamic models have shown that moderately volatile rock-forming elements such as Na, K, Si, Mg, Fe, Ca, etc. are outgassed from the magmatic surface (Schaefer & Fegley, 2007; Fegley et al., 2016; Young et al., 2019; Sossi et al., 2019). The amount of degassing volatiles from the magma ocean strongly depends on its temperature and oxygen fugacity (e.g. Sossi et al., 2019), but also on its bulk composition and solidification (Elkins-Tanton, 2008, 2012), which is interconnected with the building blocks of the planetary embryos. Even though recent studies by Hin et al. (2017) and Young et al. (2019) investigated the loss of silicates from evaporating magma oceans around low-mass planetesimals and embryos, until now no study investigated the possible escape of such elements via EUV-driven hydrodynamic drag of catastrophically outgassed steam atmosphere related H-atoms from larger planetary embryos after the disk dissipated and the magma ocean solidified.

It is very likely that processes such as atmospheric escape and collisional erosion altered the bulk composition of the Earth **from its initial stages of accretion, which is often assumed to be chondritic, to the present-day**. Earth's Si/Mg ratio for instance may have evolved due to stepwise losses from the accreting planetary building blocks (Fegley et al., 2016), **or the elevated Mn/Na ratio of small telluric bodies relative to chondrites might reflect the oxidised conditions of a magma ocean phase since Na becomes more volatile under oxidised environments than Mn (e.g. O'Neill & Palme, 2008; Siebert et al., 2018)**. The different bulk compositions due to losses of Mg, Si etc. through escape of early-formed steam atmospheres alter the density, and interior structure of planetary embryos. The

losses of Na, K, Si, Al, Ca, from crusts may further alter the composition and structure of planetary embryos and accreting protoplanets. The abundance of the isotope ^{40}K , for instance, is important since its radioactive decay contributes
95 to the thermal evolution of the interiors of planetary embryos and protoplanets and global processes such as the generation of long-lived magnetic dynamos (Turcotte & Schubert, 2002; Murthy et al., 2003; Nimmo et al., 2004; Nimmo & Kleine, 2015). In addition, elemental ratios such as Fe/Mg and Si/Fe, which can also be altered by collisional erosion (O’Neill & Palme, 2008; Carter et al., 2015; 100 Bonsor et al., 2015; Boujibar et al., 2015), might also be affected by escape of an early steam atmosphere (Fegley et al., 2016). Moreover, moderately volatile elements can be fractionated from each other through their loss from planetary embryos in dependence of their equilibrium pressure (Sossi et al., 2019). **In case of collisional erosion, elements will be fractionated according to their**
105 **incompatibility in mantle minerals during melting, while atmospheric loss preferentially removes volatile elements.**

Fig. 1 illustrates this process. Planetary embryos will experience a magma ocean phase during their accretion due to radioactive heating, gravitational energy and frequent impacts. Due to the high magma ocean temperature moderately volatile elements will outgas in dependence of the volatility and equilibrium
110 pressure and form a silicate vapor atmosphere. After dissipation of the solar nebula this atmosphere will be susceptible to atmospheric escape, for low mass bodies due to the high surface temperature and, for more massive bodies, also due the strong EUV flux from the young Sun. While small embryos might lose
115 these volatiles, bigger embryos might accumulate a silicate atmosphere **owing to their greater gravitational pull**. When the magma ocean starts to solidify a steam atmosphere – in dependence of the oxygen fugacity mainly consisting either of H_2O and CO_2 (oxidized magma ocean) or H_2 and CO (reduced magma ocean) – will start to catastrophically outgas and the silicate fraction will be
120 mixed with the outgassed volatiles. Due to the high EUV flux of the young star/Sun, H_2O and H_2 will be dissociated and atomic hydrogen dominates the upper atmosphere and will escape from the planetary embryo hydrodynamically.

cally. Heavier elements such as the moderately volatile rock-forming elements and noble gases – thereafter also called ‘trace elements’ – that reach the hydrogen dominated thermosphere will then be dragged along with the escaping hydrogen in dependence of their different masses, thus, be lost from the embryo.

Below the escaping steam atmosphere a shallow magma ocean might remain due to frequent smaller impactors which will also lead to further outgassing of moderately volatiles to keep the respective equilibrium pressures of the different escaping species. If no shallow magma ocean remains the escaping steam atmosphere will either condense or be lost completely before condensation can occur, which in turn is dependent on the orbital location. Depending on the escape rates – which are linked to the mass, surface temperature, EUV flux, and condensation time – this process will then deplete moderately volatile elements and can potentially also fractionate their isotopes and/or lighter from heavier elements. How and if this process affects depletion and fractionation of the outgassed elements on planetary embryos, however, will be the main aim of this study.

Therefore, we investigate the losses of the rock-forming elements Na, K, Si, and Mg and of the non-radiogenic noble gas isotopes ^{36}Ar , ^{38}Ar , ^{20}Ne , ^{22}Ne from planetary embryos with $0.01 M_{\text{Earth}}$ ($1.0 M_{\text{Moon}}$), $0.05 M_{\text{Earth}}$ ($0.5 M_{\text{Mars}}$), $0.1 M_{\text{Earth}}$ ($1.0 M_{\text{Mars}}$) and $0.15 M_{\text{Earth}}$ ($1.5 M_{\text{Mars}}$), hereafter referred to as $M_{\text{Emb}} = 0.5 - 1.5 M_{\text{Mars}}$, in orbit locations at 0.7, 1.0 and 1.5 AU exposed to the EUV flux of a young solar-like G-star after the disk disappeared.

To study these losses, we assume

- catastrophically outgassed steam atmospheres from a solidifying magma ocean that formed on the planetary embryos due to radioactive decay, gravitational energy, and frequent impactors during their growth;
- different degassed amounts of CO_2 and H_2O that will subsequently be lost through hydrodynamic escape;
- degassing of Na, K, Si, and Mg, as well as ^{36}Ar , ^{38}Ar , ^{20}Ne , ^{22}Ne (the ‘trace elements’) into the steam atmosphere that can then be dragged

away from the upper atmosphere by the dissociation products of H₂O;

- and that a shallow magma ocean remains beneath the steam atmosphere due to frequent impacts. If no shallow magma ocean remains, the steam atmosphere might in reality condense before it escapes completely, which strongly depends on the orbital location of the planetary embryo.

To calculate the losses of the moderately volatile elements and of the noble gases, we simulate

- the atmospheric structure of the different steam atmospheres from the surface up to the 1 μ bar-level with a 1D radiative-convective atmosphere model for planetary embryos of $M_{\text{Emb}} = 0.5 - 1.5 M_{\text{Mars}}$, magma ocean surface temperatures of $T_{\text{surf}} = 1500 - 3000$ K (Marcq, 2012; Marcq et al., 2017; Pluriel et al., 2019), and different orbital locations;
- the evolution of the steam atmosphere through energy-limited escape (e.g. Watson et al., 1981);
- and the escape of the trace elements through dragging by the hydrodynamically escaping hydrogen and oxygen atoms (Zahnle & Kasting, 1986; Hunten et al., 1987; Odert et al., 2018).

In Section 2 magma ocean parameters and the initial abundances of the studied elements as well as outgassed atmospheric fractions of the elements are discussed. Section 3 summarizes the radiative-convective magma ocean related steam atmosphere model that is used for the calculation of the atmosphere heights. In Section 4 we briefly describe hydrodynamic escape and the upper atmosphere escape model that is used for the calculation of the escape rates of the elements at different orbital locations and the therewith connected steam atmosphere parameters. We study the losses along expected solar EUV evolution tracks that correspond to slowly to moderately rotating young solar-like G-stars (Tu et al., 2015). Section 5 presents the results and discusses the findings of the study in the context of early Earth's accretion.

2. Planetary embryo composition and magma oceans

As mentioned in Section 1, Moon-mass and larger planetary embryos went through magma ocean phases as a result of efficient heating due to collisions, radioactive decay and/or gravitational energy (e.g. Elkins-Tanton, 2012; Lichtenberg et al., 2018; Albarède & Blichert-Toft, 2007). For the embryos of 0.05, 0.1, and 0.15 M_{Earth} we assume magma ocean depths of 1000 km which is in good agreement with siderophile-element abundances and silicate partitioning data as obtained for Mars (Agee & Draper, 2004; Righter et al., 1998; Righter & Chabot, 2011; Elkins-Tanton, 2012). Such magma ocean depth covers 75 % of the entire volume of the 0.05 M_{Earth} -body and 65 % and 59 % of the 0.1 M_{Earth} and 0.15 M_{Earth} -bodies, respectively.

According to Elkins-Tanton (2008, 2012), Lebrun et al. (2013), Hamano et al. (2013) and Massol et al. (2016) the solidification of magma oceans is a relatively fast process and occurs within $\approx 10^5$ years for 1.5 AU, ≈ 1.5 Myr at 1 AU, and ≥ 10 Myr at 0.7 AU, respectively. A shallow magma ocean, however, can remain after the solidification of the deep global magma ocean. A study by Maindl et al. (2015) for **instance** indicates that due to frequent impacts a shallow magma ocean might have been sustained for several million years. This is also supported by a recent study of the solidification of the martian magma ocean (Bouvier et al., 2018). This investigation of the U-Pb chronology of Martian zircons indeed indicates that the first Martian protocrust formed at about 20 Myr after the formation of the Solar System. Mars itself has accreted a mass of 0.63 M_{Mars} after ≈ 2.4 Myr and completely formed after ≈ 10 Myr (Dauphas & Pourmand, 2011), meaning that the fully formed Mars had a protracted magma ocean for about 10 Myr. **This subsequently implies that atmospheric escape can certainly play an important role in volatile depletion on a planet with a protracted magma ocean phase, since the extent of atmospheric loss on a planets volatile budget is the integral of the effective escape time.**

We assume for the Moon-mass planetary embryos magma oceans depths as derived from Lunar studies. The picritic glasses, mare basalts and potassium

Table 1: CC composition of relevant elements and their isotopes (based on Lodders (2003); Lodders et al. (2009)).

noble gases	amount		elements	amount	
	[ppm]	[mol/g]		[ppm]	[mol/g]
Ne	1.8×10^{-4}	8.92×10^{-12}	Na	4990	2.17×10^{-4}
^{20}Ne	1.60×10^{-4}	8.30×10^{-12}	Mg	95800	3.94×10^{-3}
^{22}Ne	1.20×10^{-5}	6.10×10^{-13}	Si	107000	3.81×10^{-3}
Ar	1.33×10^{-3}	3.33×10^{-11}	K	544	1.40×10^{-5}
^{36}Ar	1.13×10^{-3}	2.82×10^{-11}	Fe	185×10^3	3.32×10^{-3}
^{38}Ar	2.04×10^{-4}	5.12×10^{-12}	U	8.10×10^{-3}	3.40×10^{-11}

(K), rare-Earth elements (REE) and phosphorus (P), as well as the so-called KREEP component of enriched incompatible elements, are consistent with fractional solidification of a lunar magma ocean that was originally hundreds of kilometers deep (Philpotts & Schnetzler, 1970; Wakita & Schmitt, 1970; Elkins-Tanton, 2012). Furthermore, from observations of Lunar serial magmatism and of a uniform orthopyroxene, olivine melting region, it is expected that the lunar magma ocean depth was between 250 - 500 km (Elkins-Tanton et al., 2011; Schaefer & Elkins-Tanton, 2018). By assuming a similar size and mass for the smallest planetary embryos in our study we assume comparable magma ocean depths of 500 km for $0.01M_{\text{Earth}}$ (Moon-mass) embryos. We investigate the losses of the rock-forming elements from these Moon-mass embryos by assuming magma ocean surface temperatures (T_{Surf}) of 1500, 2000, 2500 and 3000 K.

After the disk evaporated, depending on the accretion history and orbit location, a range of possible bulk compositions for planetary embryos and accreting proto-planets can be expected. Such bodies contain a mixed rocky bulk composition with iron metal meteorites, different water abundances, or mixtures of metallic iron, silicate rock, and unconstrained volumes of icy planetesimals and planetary embryos from beyond the ice line (e.g. Brasser, 2013; Morbidelli & Raymond, 2016; Raymond et al., 2014). Based on studies of meteorites,

230 asteroids, and comets a wide range of initial water inventories in the bulk silicate composition of terrestrial planets can be expected (Elkins-Tanton, 2008, 2012; Massol et al., 2016). While we vary the outgassed water inventory in this study, we assume only a pure carbonaceous chondritic (CC) composition for all planetary embryos since these are the most volatile-rich building blocks. **Even**
235 **though our obtained results might also be valid for other bodies such as enstatite chondrites, one has to extrapolate with care. Since enstatite chondrites for instance are more reduced than carbonaceous chondrites (e.g Schaefer & Fegley, 2017), more reduced gases are also expected to be formed during their evaporation, which might alter**
240 **our results.**

Besides the rock-forming elements (Schaefer & Fegley, 2007; Schaefer & Fegley, 2010; Fegley et al., 2016) and noble gases that outgas from the magma ocean, H₂O and CO₂ molecules will degas catastrophically into dense steam atmospheres during the solidification process (Elkins-Tanton, 2008, 2012; Massol
245 et al., 2016; Salvador et al., 2017). For our simulations we assume partial pressure values of 50 bar, 100 bar and 150 bar for H₂O and 5 bar, 10 bar and 30 bar for CO₂, which are within an expected reasonable range of volatile contents for such bodies (Elkins-Tanton, 2008, 2012). Elemental and isotopic abundances relevant to our study are calculated via Lodders (2003); Lodders et al. (2009)
250 and listed in Table 2. We furthermore study different cases of outgassed amounts of the trace elements, i.e. 10%, 50% and 100% of the initial amount within the magma ocean, as well as a continuous outgassing of the studied elements in balance with the pressure equilibria between magma ocean and atmosphere.

3. Steam atmosphere and escape simulations

255 3.1. Magma ocean related radiative-convective atmospheres

A 1D radiative-convective atmosphere model is applied to the magma ocean related steam atmosphere cases discussed in Sect. 2. The model that we use is described extensively in Marcq (2012), Marcq et al. (2017), and Pluriel et al.

(2019), and applied in several coupled magma ocean-atmosphere evolution studies (Lebrun et al., 2013; Salvador et al., 2017). It takes as inputs the surface temperature and partial pressures of H₂O and CO₂. These inputs are used to compute a temperature profile, as well as density profiles for each **gas species**. Radiative fluxes are then computed at the top of the model, yielding: (i) the altitude of the 0.1 Pa (1 μ bar) level (topmost layer); (ii) the spectral reflectance and thermal radiation low-resolution spectra from 0 to 3.5×10^5 cm⁻¹.

3.2. Vertical temperature profiles

The temperature profile is prescribed in three distinct layers. From bottom to top, these are *dry adiabat*, *moist adiabat*, *isothermal*. **No ideal gas assumption is made, especially for the major component H₂O (which is a triatomic molecule).** The dry adiabatic lapse rate dT/dP follows the expression from Kasting (1988) and is valid for an H₂O (non ideal) – CO₂ (ideal) mixture. Under such conditions, the molecule H₂O behaves as given in the steam tables by Haar et al. (1984). If H₂O saturation pressure is reached, the profile switches to a moist adiabat, following also the expression given in Kasting (1988) – if H₂O saturation occurs at surface level, then there is no dry adiabat and excess H₂O is assumed to condense into a liquid water ocean; if saturation is never reached, then there is no moist adiabat layer. Finally, when the temperature reaches a specified topmost value T_0 (we adopt $T_0 = 200$ K corresponding to an average equilibrium temperature at the studied orbital distances), the profile switches to an isothermal profile at T_0 . In principle, T_0 could be computed iteratively from a null divergence of visible and thermal radiative fluxes in the topmost layer, but there are numerical stability issues which prevented us from doing so.

The model uses pressure from the surface to 0.1 Pa (1 μ bar) as a vertical coordinate. Corresponding altitudes are computed assuming hydrostatic equilibrium, following the ideal gas law for CO₂ and the steam tables of Haar et al. (1984) for H₂O, since we can easily reach a supercritical phase from H₂O. **This means that in the lowermost atmospheric layers, we may have**

$P_{\text{H}_2\text{O}} > P_{\text{crit-H}_2\text{O}}$ and/or $T > T_{\text{crit-H}_2\text{O}}$. In such a case, we cannot (and
 290 do not) consider H_2O as an ideal gas, so we use the values given in the
 aforementioned steam tables for a more accurate equation of state for
 H_2O . The pressure and temperature profiles for selected steam atmospheres up
 to the lower thermosphere, i.e. up to the $1 \mu\text{bar}$ -level, are shown in Fig. 2. The
 EUV flux, which drives the escape of the hydrogen atoms, is absorbed above
 295 this height.

Finally, the humidity ratio $\alpha_v = \rho_{\text{H}_2\text{O}}/\rho_{\text{CO}_2}$ is prescribed in the three layers
 as follows: vertically uniform in the dry adiabat and isothermal layer, and
 decreasing with increasing height in the moist adiabat layer according to the
 expression given by Kasting (1988).

300 3.3. Radiative modelling

Rayleigh scattering is considered for both H_2O and CO_2 following a simple
 λ^{-4} spectral dependency (Kopparapu et al., 2013; Snee & Ubachs, 2005).
 Gaseous absorption includes spectral lines from HITRAN 2012 (Turbet et al.,
 2017) as well as continuum opacities for $\text{CO}_2\text{-CO}_2$ (Stefani et al., 2013), $\text{H}_2\text{O-}$
 305 H_2O (Mlawer et al., 2012) and $\text{CO}_2\text{-H}_2\text{O}$ (Turbet et al., 2017). These absorption
 spectra are then used to compute a look-up table of 16 k -coefficients for a grid in
 (α_v, T, P) using KSPECTRUM (Eymet et al., 2009), to be used in a k -correlated
 code. The model then solves the radiative transfer equation using the standard
 DISORT solver in the 4-stream approximation.

310 3.4. Steam atmospheres & hydrodynamic escape

On present Earth, Venus or Mars, hydrogen escape is limited by the upward
 diffusion of hydrogen through the upper atmosphere (e.g. Hunten, 1975). The
 diffusion-limited flux is proportional to the total hydrogen-mixing ratio at the
 homopause. However, the present H_2O mixing ratio at the homopause level of
 315 these planets is quite low. In present-day Mars' case, for instance, the H_2O
 mixing ratio at the homopause level is $(3 \pm 1) \times 10^{-5}$ (e.g. Krasnopolsky &
 Feldman, 2001). As it was shown by Kasting & Pollack (1983) when they

studied the loss of water from Venus, which has an even lower H₂O mixing ratio at the homopause level of $\approx 5 \times 10^{-7}$, for levels of about $\geq 5 \times 10^{-4}$ the solar EUV energy that is absorbed at higher altitudes becomes larger than the energy that is carried away by the Jeans escape flux. Depending on the orbital location of a planetary body, its mass and the host star's energy input into the upper atmosphere from the EUV range, the atmosphere can experience classical Jeans escape or hydrodynamic outflow, which can even result in blow-off (e.g. Parker, 1963; Chamberlain, 1963; Öpik, 1963; Watson et al., 1981; Bauer & Lammer, 2004; Tian et al., 2008a,b; Erkaev et al., 2013; Lammer et al., 2013).

Jeans escape is valid as long as the upper atmosphere remains in hydrostatic conditions so that particles that populate the high-energy tail of a Maxwellian distribution at the exobase level, where the mean free path equals the scale height, overcome the escape energy and are lost from the gravitational potential of the planetary body. For H₂O mixing ratios $> 5 \times 10^{-4}$ the upper atmosphere becomes unstable against expansion and the Jeans escape formula is not valid anymore (Kasting & Pollack, 1983). These authors found that if the mixing ratios reach values that are $\geq 10^{-3}$ then the hydrogen escape flux reaches values that are comparable with the EUV-restricted energy-limited escape as discussed by Watson et al. (1981). These conditions are also fulfilled for the studied steam atmospheres where we have initial H₂O mixing ratios at the homopause levels that are more or less > 0.1 .

In case a planetary embryo underwent a magma ocean scenario so that the body outgassed an H₂O-vapor-dominated steam atmosphere during its solidification, H atoms will be the dominant species in the upper atmosphere (e.g. Kasting & Pollack, 1983; Chassefière, 1996b,a; Odert et al., 2018; Lammer et al., 2013; Guo, 2019). In the upper atmosphere the solar/stellar UV- and EUV radiation dissociates H₂O molecules and produces H atoms via the reactions $\text{H}_2\text{O} + h\nu \rightarrow \text{OH} + \text{H}$ and $\text{OH} + \text{H}_2 \rightarrow \text{H}_2\text{O} + \text{H}$. The decomposition of H₂O through these reactions is **the dominant process** (Huebner & Mukherjee, 2015; Guo, 2019). If the H₂O-vapour mixing ratio near **the homopause level** remains more or less constant for different EUV and UV fluxes, then less

H₂O-vapour will populate the altitudes above for high radiation fluxes due to
350 more efficient photodissociation and ionization. Even for present Earth, the
H₂ number density above the dissociation level is one order of magnitude lower
compared to atomic hydrogen (Lammer et al., 2013).

Recently, Guo (2019) studied the effect of photoionization and the loss of
H₂O from steam atmospheres of Earth-like planets that were exposed to EUV
355 fluxes between 10 and 400 and FUV flux values between 5 and 200 times the
present-day solar values with a hydrodynamic model that included photochem-
istry. It was found that in all cases the water molecules dissociate, around the
1 μ bar atmospheric pressure level so that atomic **H becomes the predominant
species, while O atoms are the second most abundant** (see Fig. 1; Guo,
360 2019). IR-cooling molecules such as CO₂ do not reach these altitudes largely
due to its weight, therefore the EUV flux is absorbed in a hydrogen-dominated
upper atmosphere. The hydrogen escape rate depends on the magnitude of the
stellar/solar EUV flux that drives the upper atmosphere heating and expansion
and as long as hydrogen is the main species to a lesser extent, on the oxygen
365 and CO₂ content of the atmosphere (Kasting & Pollack, 1983; Guo, 2019).

In all steam atmosphere cases that are considered in our study the solar EUV
flux is absorbed above the 1 μ bar level, where it deposits its energy that drives
the expansion of the upper atmosphere dynamically accompanied by adiabatic
cooling, so that the exobase location reaches very large distances (Watson et al.,
370 1981; Tian et al., 2005, 2008a,b; Erkaev et al., 2013, 2014, 2015; Lammer et al.,
2013), which results in an efficient escape of the hydrogen inventory or even the
loss of heavy species that are outgassed from the magma ocean and dragged
away by the escaping hydrogen atoms (Zahnle & Kasting, 1986; Hunten et al.,
1987; Pepin, 2006; Odert et al., 2018; Guo, 2019) if they reach the hydrogen-
375 dominated region.

The Tables 2, 3, and 4 show the 1 μ bar levels above the embryo surface for
masses of 0.5, 1.0, and 1.5 M_{Mars} for different surface temperatures of 1500-
3000 K and orbital locations from 0.5-1.5 AU calculated with the 1D-radiative
atmosphere model for the considered planetary embryo masses, surface tem-

Table 2: Input parameters and resulting atmospheric heights above the surface used for $M_{\text{Emb}} = 0.5M_{\text{Mars}}$.

Cases	d [AU]	T_{surf} [K]	50 bar H ₂ O	100 bar H ₂ O	150 bar H ₂ O
			5 bar CO ₂	10 bar CO ₂	30 bar CO ₂
			1 μ bar level [km]		
I	0.5	1500	2210	2292	1998
		2000	3806	3943	3313
		2500	7541	7806	6060
		3000	24386	26608	14886
II	0.7	1500	2210	2292	1998
		2000	3806	3943	3313
		2500	7541	7806	6060
		3000	24386	26608	14886
III	1	1500	2210	2292	1998
		2000	3806	3943	3313
		2500	7541	7806	6060
		3000	24386	26608	14886
IV	1.5	1500	2210	2292	1998
		2000	3806	3943	3313
		2500	7541	7806	6060
		3000	24386	26608	14886

Table 3: Input parameters and resulting atmospheric heights above the surface used for $M_{\text{Emb}} = 1.0M_{\text{Mars}}$

Cases	d [AU]	T_{surf} [K]	50 bar H ₂ O	100 bar H ₂ O	150 bar H ₂ O
			5 bar CO ₂	10 bar CO ₂	30 bar CO ₂
			1 μ bar level [km]		
V	0.5	1500	1347	1385	1245
		2000	1986	2032	1809
		2500	2946	3008	2629
		3000	4477	4567	3896
VI	0.7	1500	1347	1385	1245
		2000	1986	2032	1809
		2500	2946	3008	2629
		3000	4477	4567	3896
VII	1	1500	1347	1385	1245
		2000	1986	2032	1809
		2500	2946	3008	2629
		3000	4477	4567	3896
VIII	1.5	1500	1347	1385	1245
		2000	1986	2032	1809
		2500	2946	3008	2629
		3000	4477	4567	3896

Table 4: Input parameters and resulting atmospheric heights above the surface used for $M_{\text{Emb}} = 1.5M_{\text{Mars}}$

Cases	d [AU]	T_{surf} [K]	50 bar H ₂ O	100 bar H ₂ O	150 bar H ₂ O
			5 bar CO ₂	10 bar CO ₂	30 bar CO ₂
			1 μ bar level [km]		
IX	0.5	1500	1075	1103	1001
		2000	1524	1555	1403
		2500	2136	2172	1941
		3000	3982	3027	2677
X	0.7	1500	1075	1103	1001
		2000	1524	1555	1403
		2500	2136	2172	1941
		3000	3982	3027	2677
XI	1	1500	1075	1103	1001
		2000	1524	1555	1403
		2500	2136	2172	1941
		3000	3982	3027	2677
XII	1.5	1500	1075	1103	1001
		2000	1524	1555	1403
		2500	2136	2172	1941
		3000	3982	3027	2677

380 peratures, and steam atmosphere partial pressures. For planetary embryos of
0.1 M_{Mars} , however, no table is given because these hot and low-mass bodies
cannot build up a stable steam atmosphere due to the low gravity and high
surface temperatures which lead to extreme losses of the volatiles. This process
can be compared with the so-called ‘boil-off’ escape that occurs at very close-in
385 and hot sub-Neptune-type exoplanets (e.g. Fossati et al., 2017; Owen & Wu,
2016; Lammer et al., 2016).

As discussed before, the catastrophically outgassed steam atmospheres of the
planetary embryos escape hydrodynamically powered by the high EUV flux of
the young star/Sun (Tu et al., 2015; Erkaev et al., 2015; Odert et al., 2018). For
390 the escape calculations of the hydrodynamic H-atom escape flux with atomic
O (from dissociated H₂O molecules) as the heavy major component and the
heavy minor trace species (³⁶Ar, ³⁸Ar, ²⁰Ne, ²²Ne, CO₂, Na, K, Si, Mg) that
are embedded in the outflow, we apply the same escape model and equations
based on the energy-limited approach (Watson et al., 1981; Zahnle & Kasting,
395 1986; Hunten et al., 1987; Erkaev et al., 2016; Odert et al., 2018). Here, we
assume that the outgassed rock-forming elements are able to reach the upper
thermosphere, **making our estimates likely a maximum**. Shock waves
and a turbulent mixing of the atmosphere, by impacting material for instance,
might be responsible to lift up these elements (e.g. Genda & Abe, 2003; Trigo-
400 Rodríguez & Martín-Torres, 2013).

Since we aim to study the loss of H atoms that drag away O atoms, CO₂
molecules and the above mentioned embedded trace elements, we apply a for-
malism for the calculation of the fractionation factors, which is applicable to
more than two species. To obtain the escape rates of the above mentioned
405 trace elements, we **follow Odert et al. (2018) and use** the method of Zahnle
& Kasting (1986) and Zahnle et al. (1990). We apply the improved analyti-
cal solutions given in Zahnle & Kasting (1986) and Zahnle et al. (1990) of the
general hydrodynamic multi-component equations (Hunten et al., 1987). These
equations are applicable to atmospheres with one or two major species and an
410 arbitrary amount of additional trace species (Zahnle & Kasting, 1986; Odert

et al., 2018). For the fractionation factors of the various trace species, we apply the analytical approximations given in Zahnle et al. (1990) that are based on simplifications such as subsonic flow and isothermal conditions of the multi-species hydrodynamic equations. These equations, however, provide also good
 415 approximations for transonic atmospheric escape, since the relative fluxes of the species are already determined in the subsonic region (Zahnle et al., 1990). It should be noted that the trace elements feel the drag of both main constituents, **i.e. O and H**, but they themselves cannot influence the bulk flow of these species or other trace elements due to their low abundance.

420 The hydrogen escape flux F_H is described for a multi-component atmosphere based on the energy-limited approach by (Zahnle & Kasting, 1986; Hunten et al., 1987; Zahnle et al., 1990; Odert et al., 2018), i.e.

$$F_H = \frac{\beta^2 \eta F_{\text{EUV}}}{4\Delta\Phi(m_H + m_O f_O x_O + \sum_{i=1}^n m_i f_i x_i)}. \quad (1)$$

Here, m_H is the mass of hydrogen, m_O of oxygen, m_i of the trace elements (that is, the rock-forming elements and the Ar and Ne isotopes in this case),
 425 $\eta \approx 15\%$ (Shematovich et al., 2014) is the heating efficiency in a hydrogen dominated atmosphere, F_{EUV} is the stellar EUV flux at the orbit location, $\Delta\Phi$ is the gravitational potential at the surface of the planetary embryo. We define β as the ratio of the atmospheric planetocentric radius at the $1\mu\text{bar}$ level, the distance where the H_2O molecules are dissociated and atomic hydrogen becomes
 430 the dominant species, to the embryo surface radius. The $1\mu\text{bar}$ level, which is illustrated in Fig. 3, hence constitutes the upper atmospheric boundary in our simulation.

The mixing ratios f_i and the corresponding fractionation factors x_O and x_i are time dependent and can be written after Zahnle et al. (1990) and Odert
 435 et al. (2018)

$$x_O = 1 - \frac{g(\Delta m_{\text{O,H}}) b_{\text{H,O}}}{F_H k_B T (1 + f_O)}, \quad (2)$$

and

$$x_i = \frac{1 - \frac{g(\Delta m_{i,H})b_{H,i}}{F_H k_B T} + \frac{b_{H,i}}{b_{H,O}} f_O (1 - x_O) + \frac{b_{H,i}}{b_{O,i}} f_O x_O}{1 + \frac{b_{H,i}}{b_{O,i}} f_O}, \quad (3)$$

with $\Delta m_{O,H}$ and $\Delta m_{i,H}$ the mass difference between the heavy species and H atoms, g is the gravitational acceleration at the base of the flow, i.e. the $1\mu\text{bar}$ level, k_B is the Boltzmann constant, T the upper atmosphere temperature (**not**
 440 **to be confused with T_{surf} from Tables 2 and 3**). We use the binary diffusion coefficients of O, CO₂, Ar, Ne within hydrogen and oxygen from Zahnle & Kasting (1986). For the other trace species i the binary diffusion coefficients b are derived from the Chapman-Enskog approximation (Chapman and Cowling, 1970; Mason and Marrero, 1970) and can be written as

$$b_{j,i} \approx C \sqrt{\frac{T}{\mu_{j,i}}} \quad (4)$$

445 where the index j corresponds to the main species H and later O, while i corresponds to the trace species and $\mu_{j,i}$ is the reduced mass of the involved particles. The factor C is $\approx 1.52 \times 10^{18}$ and related to the reciprocal value of the hard sphere collision cross section (Hunten, 1973). Because $\mu_{j,i} = m_j m_i / (m_j + m_i)$, the binary diffusion coefficients $b_{j,i}$ in units of $\text{cm}^{-1} \text{s}^{-1}$ for the other trace
 450 species i within hydrogen and oxygen can then be estimated from the relation given in Koskinen et al. (2013), i.e.

$$b_{j,i} = 1.52 \times 10^{18} \sqrt{\frac{1}{m_j} + \frac{1}{m_i}} T^{0.5}. \quad (5)$$

A detailed analysis of the EUV-evolution tracks of young Sun-like stars was carried out by Tu et al. (2015). For stars younger than about 2 Gyr, the possible range of rotation related high-energy emission is large. Here we investigate
 455 the atmospheric escape for so-called slow and moderate rotators or weakly and moderately active young G-stars along their EUV evolution tracks. As discussed in Tu et al. (2015), at very young ages the EUV emission is saturated at a roughly constant fraction of the star's bolometric luminosity $L_{\text{EUV}} \sim 10^{-3} L_{\text{bol}}$ (Christian & Athioudakis, 2002), for ≈ 5 Myr, 25 Myr and 225 Myr, for slow,
 460 moderate, and fast rotator tracks. We therefore do not consider fast rotators in

this study since the H₂O contents of the magma ocean related catastrophically outgassed steam atmospheres are mainly lost within the saturation phase of a moderately rotating young G-type star (Odert et al., 2018).

The EUV flux F_{EUV} can be written as $F_{\text{EUV}}=L_{\text{EUV}}/(4\pi d^2)$ with orbital
465 distances d of 0.5 AU, 0.7 AU, 1.0 AU, 1.5 AU. We calculate the EUV-driven hydrodynamic atmospheric mass loss rate over time along slow and moderate rotator EUV luminosity evolution tracks, i.e. $L_{\text{EUV,s}}$ and $L_{\text{EUV,m}}$, respectively, based on the rotation rate distribution of young solar-like stars according to Tu et al. (2015), i.e.

$$L_{\text{EUV,s}} = 5.75 \times 10^{31} t^{-0.93} \quad (6)$$

$$L_{\text{EUV,m}} = 4.70 \times 10^{32} t^{-1.18}, \quad (7)$$

470 where t is the stellar age in Myr and L_{EUV} the EUV luminosity in erg/s. The EUV flux, the mixing ratios and the fractionation factors are therefore calculated as functions of time. This provides the description of the evolving elemental masses for the considered isotope and elemental ratios.

Recently the solar nebula life time was estimated to be around 3.3-4.5 Myr
475 constrained by meteorite paleomagnetism (Wang et al., 2017) and inferred from Pb isotopic ages of chondrules (Bollard et al., 2017). Further, the solidification of the deep magma ocean together with the catastrophically outgassing of the steam atmosphere is assumed to take place within a few hundred thousand years (e.g. Elkins-Tanton, 2008, 2012). Therefore ≈ 5 Myr seems to be a reasonable
480 starting time.

In the following section we present the results of the atmospheric escape model and discuss the implications of the dragged and lost rock-forming elements from dissociated planetary embryos for terrestrial planet evolution.

4. Results

4.1. Planetary embryos between $0.5 M_{\text{Mars}}$ and $1.5 M_{\text{Mars}}$

The loss and evolution of steam atmospheres from planetary embryos at 1 AU for a slowly rotating young G-star (cases III, VII and XI as described in Tables 2, 3 and 4) for $M_{\text{Emb}} = 0.5, 1.0$ and $1.5 M_{\text{Mars}}$ and for a surface temperature of 1500 K is shown in Fig. 4. Even though at 1 AU the condensation time of a magma ocean is estimated to be around 1.5 Myr (see (e.g. Lebrun et al., 2013; Massol et al., 2016), here, it is assumed that a shallow magma ocean remains after the solidification of their deep global magma ocean (see also Section 2). If no shallow magma ocean can be sustained, however, the hydrodynamic escape will be reduced or even stop after the steam atmosphere condensed. In general, for scenarios in which the steam atmosphere is lost within ≈ 10 Myr at 0.7 AU, ≈ 1.5 Myr at 1 AU, and ≈ 0.1 Myr at 1.5 AU no shallow magma ocean will be needed due to the condensation time (Lebrun et al., 2013) being longer than the duration of the total escape of the steam atmosphere.

One can see in Fig. 4 that the steam atmospheres for these embryos are lost within ≈ 0.9 Myr (i.e. 5.9 Myr after formation of the Solar System if one assumes that escape starts after 5 Myr) for $0.5 M_{\text{Mars}}$, ≈ 5.5 Myr for $1.0 M_{\text{Mars}}$, and within ≈ 16.7 Myr for $1.5 M_{\text{Mars}}$ respectively. For cases with masses $M < 1.5 M_{\text{Mars}}$ even CO_2 molecules will be completely lost via the hydrodynamic drag of hydrogen which is in agreement with previous studies (Erkaev et al., 2014; Odert et al., 2018). For the particular case with $M = 1.5 M_{\text{Mars}}$ in Fig. 4 a small amount of CO_2 and O might remain, but will likely be rapidly lost due to other escape processes such as subsequent thermal escape of C and O (e.g. Tian et al., 2009) or ion-escape (e.g. Dong et al., 2018).

It has to be noted that for moderate and fast rotating young solar-like stars but also for hotter surface temperatures, the atmospheres are lost even more rapidly as can be seen in Fig. 5. Here, this Figure illustrates the time of the escape of the whole atmosphere of planetary embryos with $M_{\text{Emb}} = 0.5, 1.0$ and $1.5 M_{\text{Mars}}$ in dependence of their orbit and surface temperature for a slow

and moderate rotating young G-star. As one can see for $0.5 M_{\text{Mars}}$ in 1 AU
515 most of the steam atmosphere will be lost within the formation time of the
Martian crust; in general, if one assumes similar formation times for the first
protocrust **of planetary embryos** as for Mars (Bouvier et al., 2018) then
almost all of these atmospheres can be lost even for a protoplanetary mass of
 $1.0 M_{\text{Mars}}$ up ≈ 1.5 AU. For $1.5 M_{\text{Mars}}$ only for a moderate rotator most of the
520 steam atmospheres can be lost within this time frame. Additionally, one should
also note that also for more reduced conditions, i.e. a magma ocean which is
not oxidized and, therefore, preferentially outgasses H_2 and CO , atmospheres
might be lost more rapidly than in our simulations (Odert et al., 2018).

Figures 6, 7, and 8 show the evolution of the ratios of several rock-forming
525 elements and noble gases for the total abundance of the respective planetary
embryo (i.e. total abundance within the atmosphere, magma ocean and rest-
mass of the embryo) as mentioned above (see Table 2 for the same cases as in
Fig. 4 but for all surface temperatures). These figures correspond to cases in
which we assumed 100% of the trace elements to be in the steam atmosphere.
530 Our model approach showed that, regardless of the initial value, almost the
whole amount of the trace elements within the escaping steam atmosphere will
be lost in any cases. The results will furthermore be nearly identical regardless
whether we assume continuous outgassing from the magma ocean or everything
to be outgassed at the beginning.

535 For the noble gases, Figures 6, 7, and 8 also display the evolution of their
individual abundances (in mol/g). It can be clearly seen that all (moderately)
volatile elements that will be outgassed from the magma ocean and further
transported to the upper thermosphere will be completely lost together with the
steam atmosphere. This also means that at such small-mass bodies noble gas
540 isotope ratios will not be significantly fractionated by this escape process but can
only be depleted. Volatile rock-forming elements and noble gases, however, will
be not only depleted but also fractionated against less volatile **but also against
refractory elements that stay in the melt such as U**. These elements
might further also be fractionated against the heavier and siderophile element

545 Fe, in particular if the planetary embryo is already differentiated and most of the iron resides in the core. For an undifferentiated body, however, it has to be taken into account that Fe can under certain conditions even become more volatile than Mg and hence also escape from the planetesimal. For $M_{\text{Emb}} \leq 0.5 M_{\text{Mars}}$ and $T_{\text{surf}} \geq 2500$ K the escape related depletion occurs within ≤ 0.2 Myr. Within
 550 the inner Solar System, such bodies will therefore always be strongly depleted in volatile elements even in the event that its magma ocean crystallises completely over a timescale shorter than that for atmospheric loss.

4.2. Moon-mass and smaller planetary embryos

For Moon-mass and smaller bodies a more simple way of estimating the
 555 escape can be **applied, provided these bodies have hot magmas oceans and sufficiently high T_{surf} , then escape should be immediate.** Due to the hot magma ocean and the low gravity of such low-mass embryos the ratio of the gravitational to thermal energy of the outgassed elements results in immediate hydrodynamic loss independent of the incident EUV flux. No
 560 dense silicate atmosphere or steam atmosphere can build up. The corresponding escape parameters, i.e.

$$\lambda = \frac{GM_{\text{emb}}m_i}{k_B T_{\text{surf}} R_{\text{emb}}} \quad (8)$$

for the different elements on the surface of a Moon-mass embryo can be found in Table 5. Here, G is the Newtonian gravitational constant, k_B the Boltzmann-constant, M_{emb} and R_{emb} are the mass and radius of the planetary embryo, and
 565 m_i are the masses of the different outgassed elements. According to Volkov et al. (2011) and Erkaev et al. (2015) the thermal escape regime changes over a narrow range of the critical escape parameter λ_{crit} ; the escape is purely hydrodynamic for $\lambda_{\text{crit}} \leq 2 - 3$ whereas for $\lambda \geq 6$ it is not. We therefore assume that for $\lambda \leq 6$ any outgassed elements are lost immediately to space. As one can see
 570 from this table, for $T_{\text{surf}} \geq 2500$ all elements **studied** except Fe are below $\lambda = 6$; for cooler magma ocean surface temperatures some of them are slightly above. In such a case, however, one can expect that these elements nevertheless experience high escape and will be ionized and picked-up by the solar wind.

Table 5: Escape parameters λ on the surface of a Moon-like planetary embryo for different rock-forming elements, iron, noble gas isotopes and for H_2O and its dissociation products as a function of the magma ocean surface temperature T_{surf} .

element	λ (1500 K)	λ (2000 K)	λ (2500 K)	λ (3000 K)
Na	5.20	3.90	3.12	2.60
Mg	5.50	4.13	3.30	2.75
Si	6.36	4.77	3.82	3.18
SiO	9.96	7.47	5.98	4.98
K	8.85	6.64	5.31	4.43
Fe	12.64	9.48	7.59	6.32
FeO	16.27	12.20	9.76	8.13
^{36}Ar	8.15	6.11	4.89	4.08
^{38}Ar	8.60	6.45	5.16	4.30
^{20}Ne	4.53	3.4	2.26	2.26
^{22}Ne	4.98	3.74	2.99	2.49
H_2	0.46	0.34	0.27	0.23
O	3.60	2.7	2.17	1.80
H_2O	4.10	3.10	2.45	2.09

575 These bodies will therefore be strongly depleted in rock-forming elements and noble gases. Table 6 shows as an example of the maximum amount of elements that can be lost from an assumed 500 km deep magma ocean which is composed out of CC-like material as shown in Table 2. In case all of these elements in the magma ocean outgas into space then the amount given in Table 6 represents the depletion of the particular elements.

580 As has been shown by several different studies (e.g. Kazenas et al., 1985; ?; ?; ?) Si will most probably outgas as a molecule, i.e. SiO. Table 5 therefore also shows λ for SiO. **For steam atmospheres, one might also consider that these elements may also form hydroxides, chlorides or sulfates, thereby increasing their molar mass and hence their λ .**

Table 6: Amount of rock-forming elements and noble gases in a 500 km deep magma ocean of a moon-like planetary embryo with 100% CC composition (see also Lodders et al. (2009)).

element	amount in magma ocean [g]	amount in magma ocean [%]
Na	1.02×10^{22}	0.499
Mg	1.85×10^{23}	9.58
Si	1.79×10^{23}	10.7
Ne	4.18×10^{14}	1.08×10^{-8}
Ar	1.56×10^{15}	1.33×10^{-7}
K	6.53×10^{20}	0.0544

585 For temperatures of $T_{surf} < 2500$ K SiO would have a λ that would be above the critical value and might therefore build up a tenuous silicate atmosphere as a consequence of an equilibrium between outgassing and escape. Such molecules, however, are also susceptible to dissociation due to the strong EUV flux. In case that the outgassing flux from the magma ocean cannot be maintained this
590 atmosphere will soon be depleted by various escape processes such as ion-pickup, sputtering and thermal escape.

Table 5 also shows Fe and FeO. As can be seen for both – the molecule and atomic iron – λ is clearly above λ_{crit} for all magma ocean surface temperatures, meaning that Fe only escapes through Jeans escape and potentially through non-
595 thermal escape processes, while Mg and Si can escape more easily. Therefore it seems likely that for planetary embryos that are not capable of keeping a stable steam atmosphere Fe might be enriched against Mg and Si, even if the body is not differentiated.

Recently Young et al. (2019) investigated the escape of a rock-vapour at-
600 mosphere with an average molecular mass of 34 g/mol that was outgassed from a planetary embryo with a radius of 700 km and 0.5 Pluto-masses (i.e. with $0.088 M_{Moon}$), and a magma ocean surface temperature of 1830 K. These authors assumed that this outgassed atmosphere obtains a hydrostatic equilibrium and builds up a stable atmosphere as an equilibrium between outgassing

605 and ‘hydrodynamic escape’ with a well mixed convective layer (troposphere),
a radiative layer (stratosphere), and a collision-less exosphere with a relative
depth controlled by the mass and surface temperature. However, under these
planetary parameters the assumption of Young et al. (2019) of a hydrostatic
equilibrium is physically not valid.

610 A hydrostatic equilibrium of an atmosphere occurs when external forces, such
as the gravity of the body, are balanced by the pressure-gradient force of the
atmosphere. By using the parameters given in Young et al. (2019), the escape
velocity on the surface of such an embryo approximately equals the mean speed
of the Maxwell-Boltzmann distribution of the bulk rock-vapor atmosphere. This
615 corresponds to a Jeans escape parameter on the surface of $\lambda = 1.38$. As already
described above, such an atmosphere is by no means in a hydrostatic equilib-
rium but clearly in a regime of strong hydrodynamic escape (e.g. Chamberlain,
1963; Bauer & Lammer, 2004; Volkov et al., 2011; Erkaev et al., 2014). Fur-
thermore, the pressure in their assumed convective layer that is in the order of
620 $\leq 2 \times 10^{-8}$ bar is less than a typical atmospheric pressure at the homopause
level, meaning that eddy diffusion is not fulfilled and the atmosphere cannot be
assumed to be well mixed; each species follows its own scale height. Under such
conditions gravity cannot prevent the atmosphere from diffusing into space and
no stable atmosphere can be maintained and the outgassing flux approximately
625 equals the escape rate.

Furthermore, we tried to simulate **a steam atmosphere** with our 1D
radiative-convective atmosphere model (see Section 3) with the same param-
eters as in the study of Young et al. (2019) for a protoplanet with a radius of
700 km, a mass of $0.5 M_{\text{Pluto}}$, and a surface temperature of 1830 K and found
630 a divergence in hydrostatic altitude above $P/P_{\text{surf}} = 0.2$ (see Fig. 9, meaning
that atmospheric build-up is impossible in such a case due to a direct escape of
the molecules at thermal or sound speed unless there is a steady supply from
the surface/interior to compensate for these losses. For a silicate atmosphere
around such bodies, this means that the entire moderately volatile elements in
635 the magmatic layer will eventually evaporate.

The assumption of a hydrostatic equilibrium in the study of Young et al. (2019) is thus only justified for cases in which the Jeans escape parameters of the outgassed elements at the surface of an embryo exceed $\lambda \approx 6$ (Erkaev et al., 2015; Volkov et al., 2011).

640 We additionally applied an adopted version of the 1D hydrodynamic atmosphere model as described in Erkaev et al. (2015) and Erkaev et al. (2016) to a similar silicate atmosphere as assumed by Young et al. (2019) with an average atomic mass of 34 amu. Here, Fig. 10a shows the evolution of the atmospheric temperature (red and
645 blue lines) and the corresponding λ (green and black) as retrieved by our model and by Young et al. (2019), respectively. In our case, the temperature is decreasing much more rapidly than in the model of Young et al. (2019) due to the adiabatic cooling of the hydrodynamically outflow of the bulk gas. On the top of the atmosphere at above
650 $\approx 9 R_{\text{body}}$ the EUV flux, which we scaled to the value expected for the early solar system after the solar nebula dissipated (Tu et al., 2015), is absorbed, heats the gas and builds up a thermosphere layer. The escaping hydrodynamic flow is supersonic just from the lower boundary for a low λ of ≈ 1.4 . From the simulation, we also see that
655 the density is decreasing rather slow, still being large at the upper boundary. Because of that, the EUV flux cannot penetrate deep into the atmosphere and, thus, it is absorbed in the upper atmospheric layers. Here, the absorbed EUV flux produces heating and a growth in temperature and pressure. The interaction between the supersonic
660 hydrodynamic flow and the heated region leads to an appearance of a standing shock. At the shock front one can see a velocity drop (see Fig. 10a) due to a density rise, but the loss rate remains constant. In case of very a low lambda (≈ 1), the interaction between the escaping supersonic flow from the atmosphere and incoming EUV flux results
665 in the appearance of a stationary standing shock wave at the top of the atmosphere.

However, concerning the equilibrium of outgassing and escape rates that converges into a particular surface pressure, the results of Young et al. (2019), even though the assumption of a hydrostatic atmosphere is physically not valid under such conditions, do not differ much from ours as shown in Fig. 11. While Young et al. (2019) obtained a pressure of $\approx 1 \times 10^{-8}$ bar and an evaporation and escape rate of $\approx 7.5 \times 10^{32} \text{ s}^{-1}$, we retrieve values of $\approx 2 \times 10^{-8}$ bar for the pressure and $\approx 5.5 \times 10^{32} \text{ s}^{-1}$ for the evaporation and escape rate. The reason for the only minor divergence can be found in the fact that such a hot and low-mass body just cannot hold the gas and everything evaporates into space, leading to similar results.

5. Discussion and conclusion

The different escape scenarios are summarized in Fig. 12. While for small planetary embryos with masses of $M \leq 1 M_{\text{Moon}}$ no dense atmosphere can build up due to the strong escape of volatile elements that are outgassed from the magma ocean, more massive embryos will build up a steam atmosphere during magma ocean solidification which will later-on escape due to strong EUV driven hydrodynamic escape. The escaping H atoms will drag away heavier elements which reach the upper atmosphere in dependence of their weight.

In Fig. 13 one can see the relation between the orbital location d of the planetary embryo, the magma ocean surface temperature and the amount of loss of K. The reason for the decrease in escape for larger orbit distances is the declining intensity of the EUV flux which decreases with $1/d^2$. For all simulated cases the **entire budget of outgassed** moderately volatile elements and noble gases from the magma ocean are lost due to hydrodynamic escape as long as the embryo is located within about $d = 2 \text{ AU}$, **which is the reason why the profiles of Fig. 13 flatten for low heliocentric distances. For a complete loss**, however, the timescale of the condensation matters which is related to the distance and the lifetime of a possible shallow magma ocean

below the steam atmosphere. If the surface cools down, the remaining H₂O vapor condenses, escape decreases and part of the volatile elements will remain at the embryo.

It has further to be noted that even though these planetary embryos can lose most of their noble gases, the initial fractionation of ²⁰Ne/²²Ne and ³⁶Ar/³⁸Ar won't change significantly.

Table 7: Escape duration for cases with an atmospheric pressure of 50 bar H₂O and 5 bar CO₂ and a T_{surf} of 1500 K for planetary embryos with masses of $0.5 M_{\text{Mars}}$ (cases I-III from Table 2), $1.0 M_{\text{Mars}}$ (cases V-VII from Table 3), and $0.5 M_{\text{Mars}}$ (cases IX-XI from Table 4) and for orbital locations d between 0.5 and 1 AU.

cases	slow rotator	moderate rotator
	time [Myr]	time [Myr]
I	0.26	0.15
II	0.54	0.29
III	1.11	0.61
V	0.77	0.42
VI	1.51	0.83
VII	3.09	1.69
IX	1.15	0.63
X	2.27	1.24
XI	4.66	2.55

The escape of rock-forming elements which has been obtained with these simulations might ultimately also alter the respective Fe/Mg and Si/Fe ratios not only of these proto-planetary embryos but also of the planets that will accrete these embryos. Carter et al. (2015) and Bonsor et al. (2015) tried to reproduce the present-day Mg/Fe and Fe/Si ratios via collisional erosion of differentiated bodies with the assumption of the Earth being accreted from chondritic material. In their simulations, however, Earth can only be reproduced with significant dynamical excitation of the growing embryos (Carter et al.,

710 2015) and/or if Earth is an extreme case and the core is assumed to contain
10% silicon by mass (Bonsor et al., 2015). Since our mechanism of thermal
escape might also partially account for the very high Fe/Mg and low Si/Fe
ratios of the present-day Earth, such extreme dynamical scenarios or core Si
abundances might therefore not be necessary. In this context, however, further
715 studies of the escape of iron combined with impact erosion simulations will be
needed in the future. Even though Fe is heavier than Mg and can, thus, less
easily escape (see also Table 5), it is on the other hand more volatile than Mg
and hence most likely more abundant in such an escaping atmosphere.

The significant loss of the radioactive heat producing **isotope** ^{40}K will take
720 place at these embryos due to these processes which should therefore also play
an important role in the development of tectonics and hence the potential habit-
ability of terrestrial planets. Extrasolar Systems with different stellar EUV evo-
lution and planetary accretion history than the Solar System might ultimately
lead to significantly different compositional abundances of heat producing ele-
725 ments in their respective planets which can affect their geophysical properties
such as their tectonic regime as well as their potential magnetic dynamo (e.g.
??). Whether such planets will end up with more or less ^{40}K as e.g. Earth
might therefore be a crucial key factor in them being habitable or not.

Since the reservoir of elements that potentially can outgas and escape de-
730 pend on the magma ocean depth, the depletion of moderately volatile elements
might change as well, if the depth of the magma ocean is different than assumed
in our study. It should, however, also be noted that the ratios and depletions
of the volatile rock-forming elements calculated in this study most likely repre-
sent an upper value for the given magma ocean depth. The two main reasons
735 for this are: i) Although the whole outgassed amount of these elements will be
lost to space, it is unclear whether 100% of the respective species will indeed
be incorporated into the steam atmosphere or if a fraction remains inside the
magma ocean. ii) The outgassed elements have to overcome a bottle neck from
the lower atmosphere to the thermosphere, i.e. the so-called cold-trap, where
740 they can reach the condensation temperature. Frequent impactors during the

early phase of the Solar System, however, might cause shock waves and turbu-
lently mix the atmosphere which can lift these elements above the cold-trap into
the thermosphere (e.g. Genda & Abe, 2003; Trigo-Rodríguez & Martín-Torres,
2013). Alternatively, one should not forget that bigger planetary embryos form
745 from the collisions of smaller ones. As discussed in Section 4.2, these building
blocks also underwent strong depletion during their boil-off phases so that one
can expect the final composition to be a mixture of these complex interactions
which is planned to be studied in more detail in the future. For noble gases,
however, the cold-trap in steam atmospheres of larger bodies is not existing due
750 to their low condensation temperatures.

From the results shown in Section 4 one can see that for the loss of ele-
ments the difference between a slow and a moderate rotator consists only in the
duration of the loss (Table 7). It doesn't matter whether the Sun was a slow,
moderate or fast rotator, planetary embryos of up to $1.5 M_{\text{Mars}}$ that are orbiting
755 in close distance to the Sun can either lose all volatiles that are outgassed from
the magma ocean (within ≈ 2 AU) or at least a significant amount of the out-
gassed material for orbital location to several AU. A growing protoplanet might
therefore accrete significantly volatile depleted material after the stellar nebula
dissipated and during the so-called giant impact phase. Our results indicate in
760 agreement with Sossi et al. (2019) and Lammer et al. (2019) that Earth's volatile
inventory is a product of the accretion of various smaller planetary embryos that
had experienced different levels of volatile escape. As Lammer et al. (2019) has
shown in agreement with Marty (2012) through analysis of the present-day Ne
and Ar isotopic ratios, Earth accreted $\approx 0.95 M_{\text{Earth}}$ from building blocks that
765 were extremely depleted in volatiles which illustrates that most of the accreted
material has indeed suffered from such extreme escape processes. A late accre-
tion of chondritic material via large impactors might therefore not necessarily
lead to a volatile rich planet.

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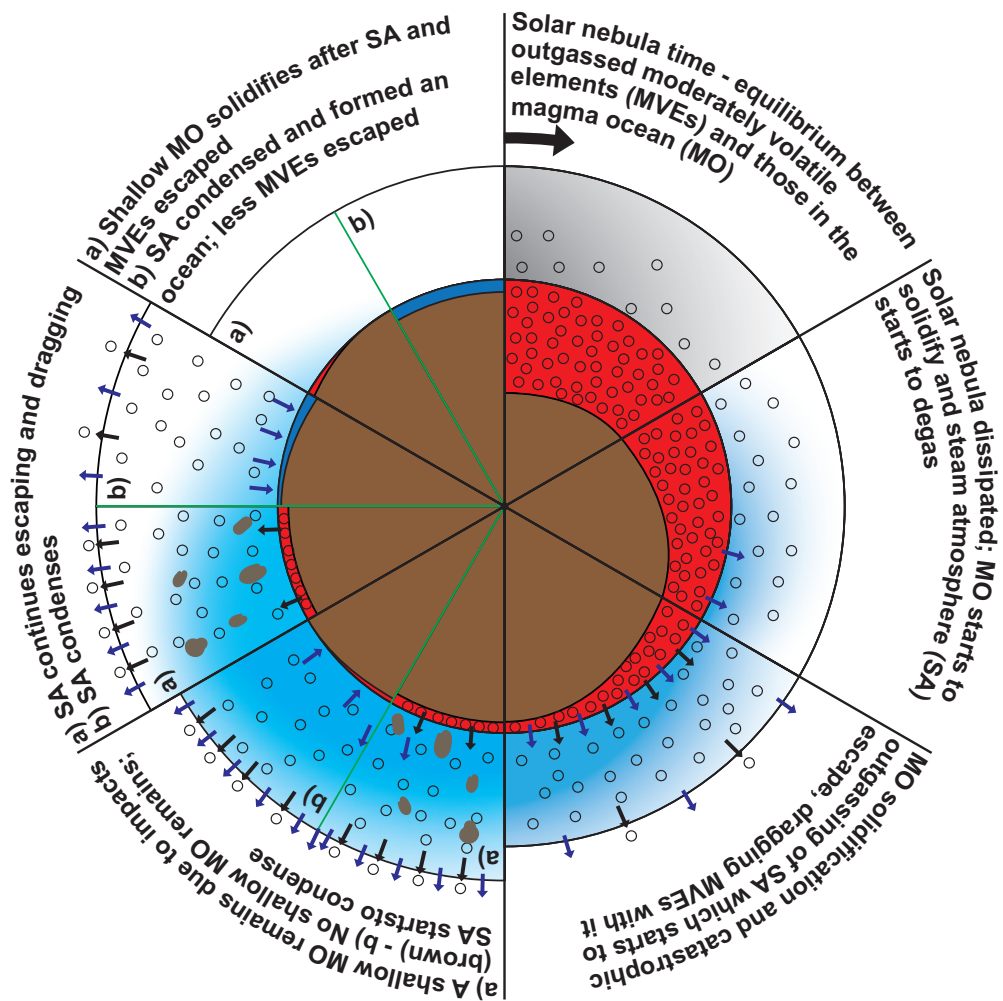


Figure 1: Illustration of the assumed evolutionary scheme from the nebula and magma ocean phase to escaping/condensing steam atmospheres.

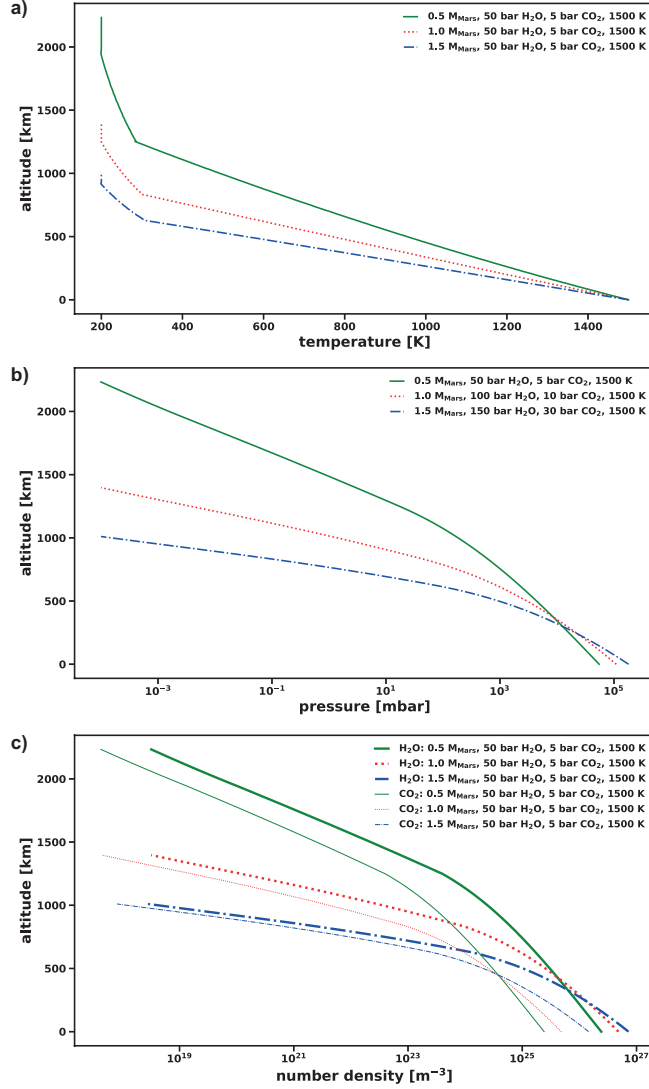


Figure 2: a) Temperature profiles for three selected steam atmospheres below the thermosphere as simulated with the 1D radiative-convective atmosphere model, i.e. for an orbital distance of 1.0 AU, $T_{\text{surf}} = 1500$ K, and for a slow rotator, i.e. $M_{\text{Emb}} = 0.5 M_{\text{Mars}}$, 50 bar H_2O and 5 bar CO_2 (Case III; upper panel), $M_{\text{Emb}} = 1.0 M_{\text{Mars}}$, 100 bar H_2O and 10 bar CO_2 (Case VII; middle panel), and $M_{\text{Emb}} = 1.5 M_{\text{Mars}}$, 150 bar H_2O and 30 bar CO_2 (Case XI; lower panel). b) Pressure profiles for the same three cases. c) **Number densities of H_2O and CO_2 for the same cases.**

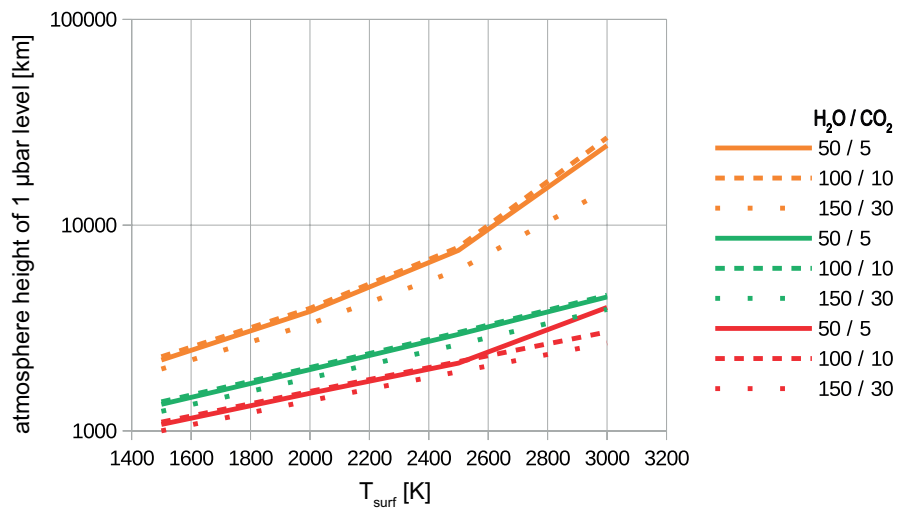


Figure 3: Relation between the surface temperature of the planetary embryo and the steam atmosphere height at the $1 \mu\text{bar}$ level for various atmospheric partial pressures ($\text{H}_2\text{O}/\text{CO}_2$). These heights depend mainly on T_{surf} and not on the stellar luminosity, i.e. these are independent from the orbital location for these particular scenarios. Orange line: $0.5 M_{\text{Mars}}$; green line: $1.0 M_{\text{Mars}}$; red line: $1.5 M_{\text{Mars}}$.

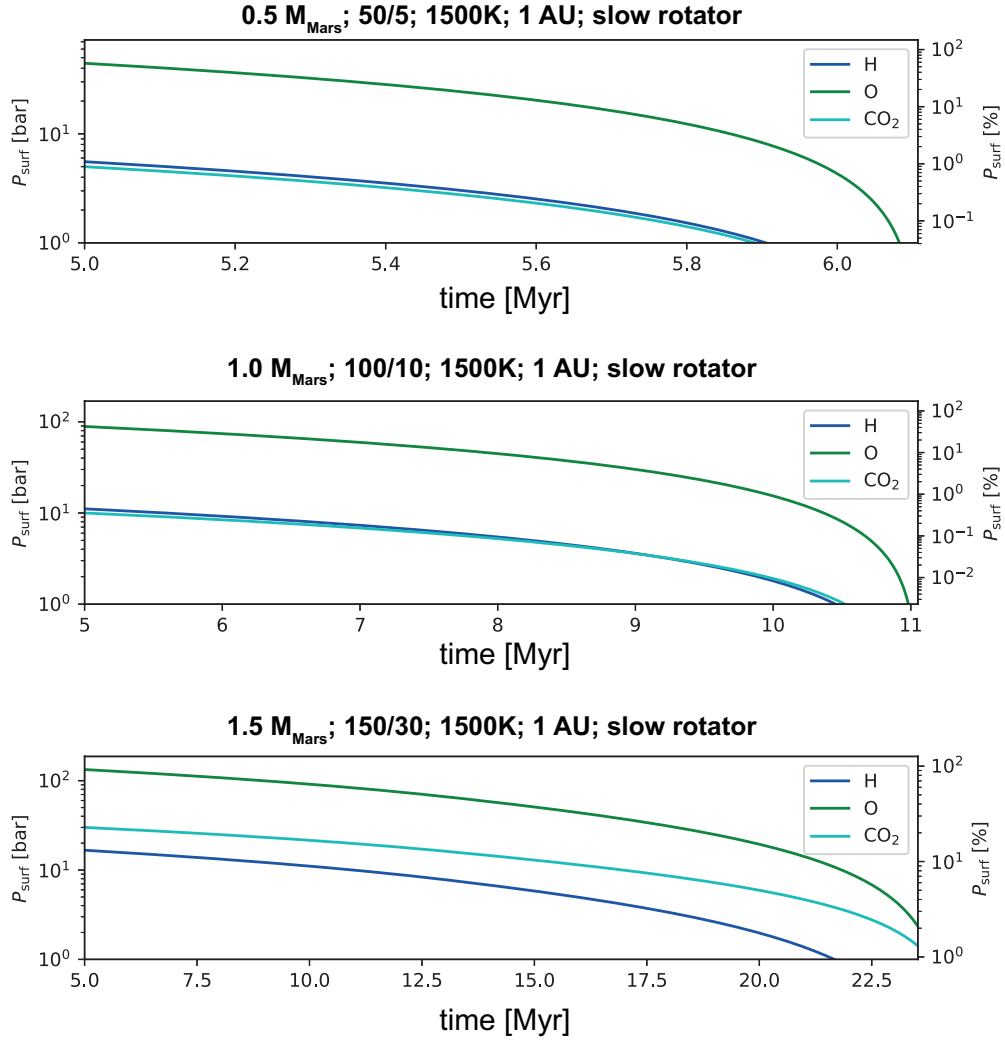


Figure 4: Atmospheric partial surface pressure evolution for different scenarios at an orbital distance of 1.0 AU, $T_{\text{surf}} = 1500$ K, and for a slow rotator, i.e. $M_{\text{Emb}} = 0.5 M_{\text{Mars}}$, 50 bar H₂O and 5 bar CO₂ (Case III; upper panel), $M_{\text{Emb}} = 1.0 M_{\text{Mars}}$, 100 bar H₂O and 10 bar CO₂ (Case VII; middle panel), and $M_{\text{Emb}} = 1.5 M_{\text{Mars}}$, 150 bar H₂O and 30 bar CO₂ (Case XI; lower panel).

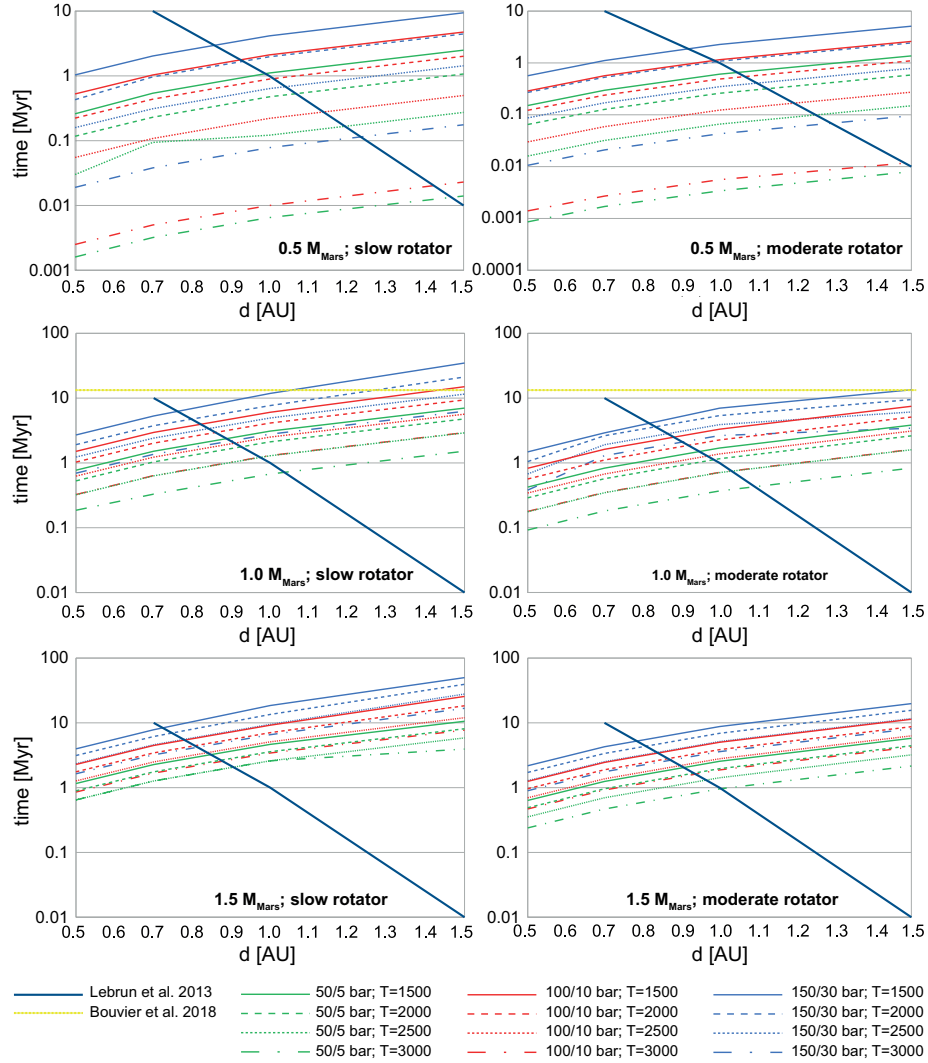


Figure 5: Duration for the total escape of the steam atmosphere for different protoplanetary masses, atmospheric pressures and temperatures for a slow and a moderate rotator in dependence of the orbital distance. The solid dark blue line in each panel shows the steam atmosphere solidification time based on the study of Lebrun et al. (2013); the dotted yellow line in the $1 M_{\text{Mars}}$ -panels show the time of the earliest protocrust at Mars (Bouvier et al., 2018).

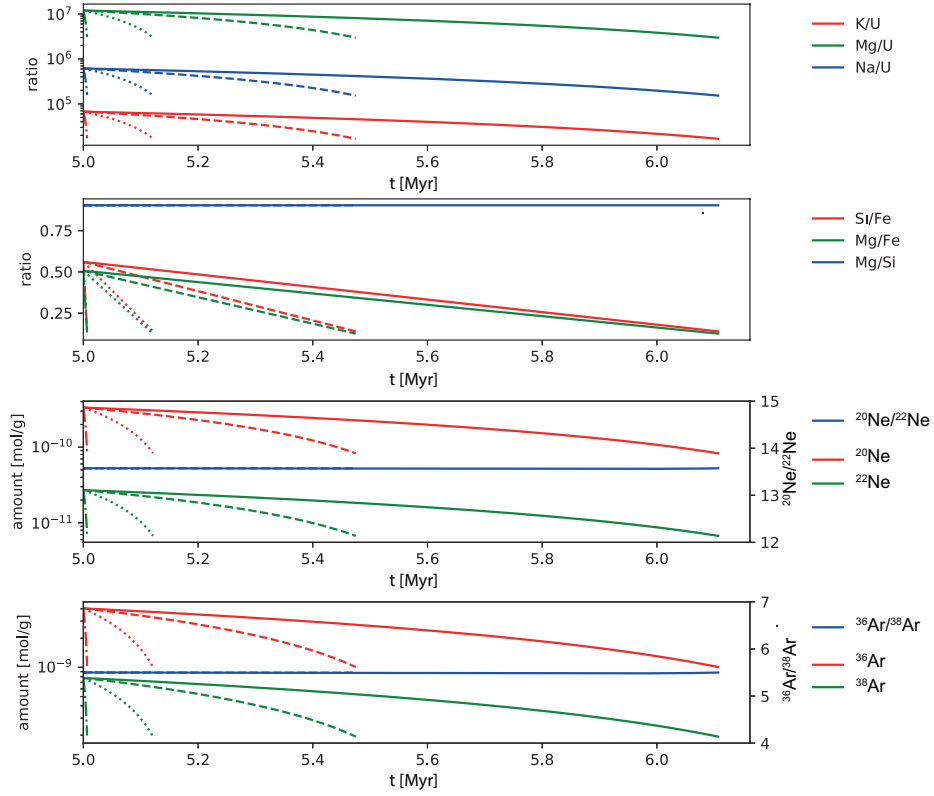


Figure 6: Results for $M_{\text{Emb}} = 0.5 M_{\text{Mars}}$, an orbital distance of 1 AU (case III), a slow rotator, 50 bar H_2O and 5 bar CO_2 . About 75% of the total amount of the planetary embryo is lost, i.e. almost everything (>90%) of the trace species that was contained in the magma ocean. Solid line: $T_{\text{surf}} = 1500$ K; dashed line: $T_{\text{surf}} = 2000$ K; dotted line: $T_{\text{surf}} = 2500$ K; dash-dotted line: $T_{\text{surf}} = 3000$ K. The different line-ends correspond to the loss-time of the steam atmosphere. The ratios are calculated for mol/g and U as well as Fe were assumed to remain at the planet, i.e. their abundance was assumed to be constant.

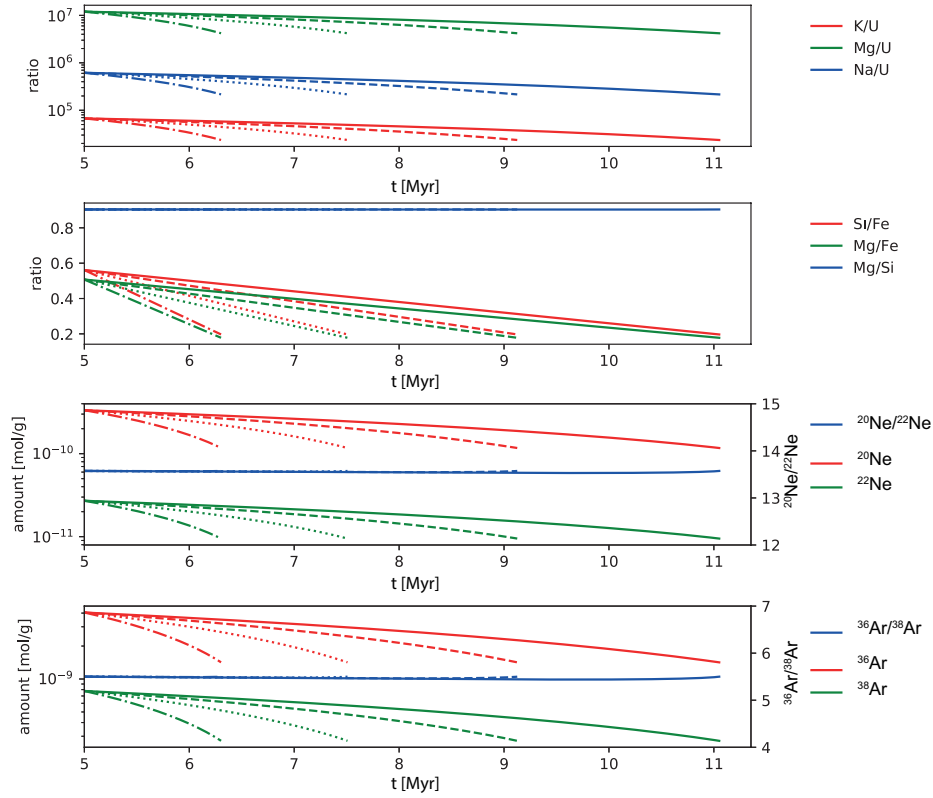


Figure 7: Results for $M_{\text{Emb}} = 1.0 M_{\text{Mars}}$, an orbital distance of 1 AU (case VII), a slow rotator, 100 bar H_2O and 10 bar CO_2 . About 65% of the total amount of the planetary embryo is lost, i.e. almost everything (>90%) of the trace species that was contained in the magma ocean. Solid line: $T_{\text{surf}} = 1500 \text{ K}$; dashed line: $T_{\text{surf}} = 2000 \text{ K}$; dotted line: $T_{\text{surf}} = 2500 \text{ K}$; dash-dotted line: $T_{\text{surf}} = 3000 \text{ K}$. The different line-ends correspond to the loss-time of the steam atmosphere. The ratios are calculated for mol/g and U as well as Fe were assumed to remain at the planet, i.e. their abundance was assumed to be constant.

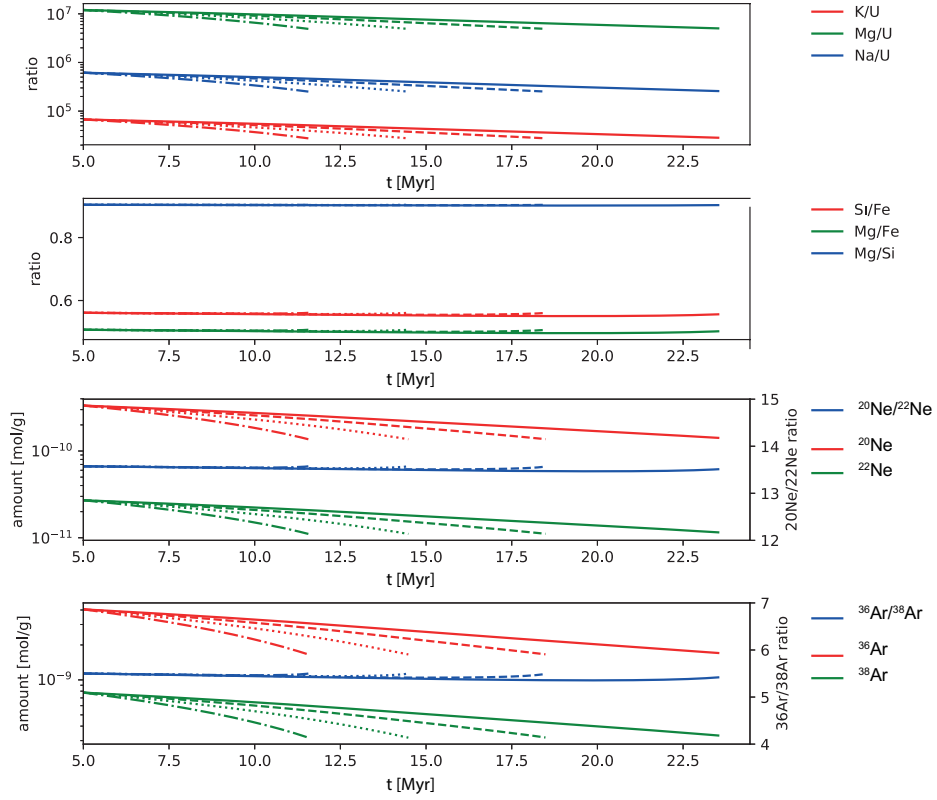


Figure 8: Results for $M_{\text{Emb}} = 1.5 M_{\text{Mars}}$, an orbital distance of 1 AU (case XI), a slow rotator, 150 bar H_2O and 30 bar CO_2 . About 58% of the total amount of the planetary embryo is lost, i.e. almost everything (>90%) of the trace species that was contained in the magma ocean. Solid line: $T_{\text{surf}} = 1500$ K; dashed line: $T_{\text{surf}} = 2000$ K; dotted line: $T_{\text{surf}} = 2500$ K; dash-dotted line: $T_{\text{surf}} = 3000$ K. The different line-ends correspond to the loss-time of the steam atmosphere. The ratios are calculated for mol/g and U as well as Fe were assumed to remain at the planet, i.e. their abundance was assumed to be constant.

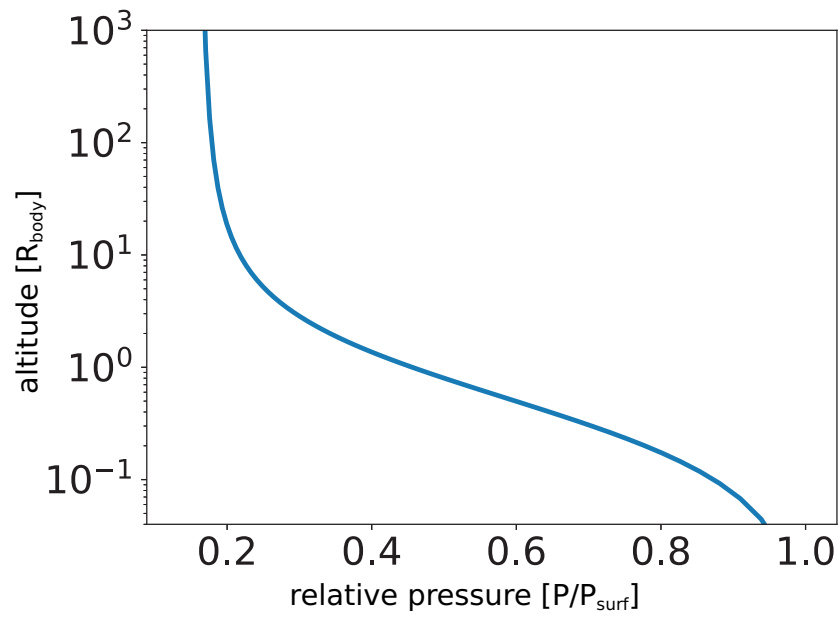


Figure 9: Simulation run with our 1D radiative-convective atmosphere model (as described in Section 3) of the case described in Young et al. (2019) with $M=0.5 M_{\text{Pluto}}$, $R=700 \text{ km}$ and $T_{\text{surf}}=700 \text{ K}$. In our model the pressure does not drop to zero as altitude goes to infinity, i.e. no stable solution was found for this particular case.

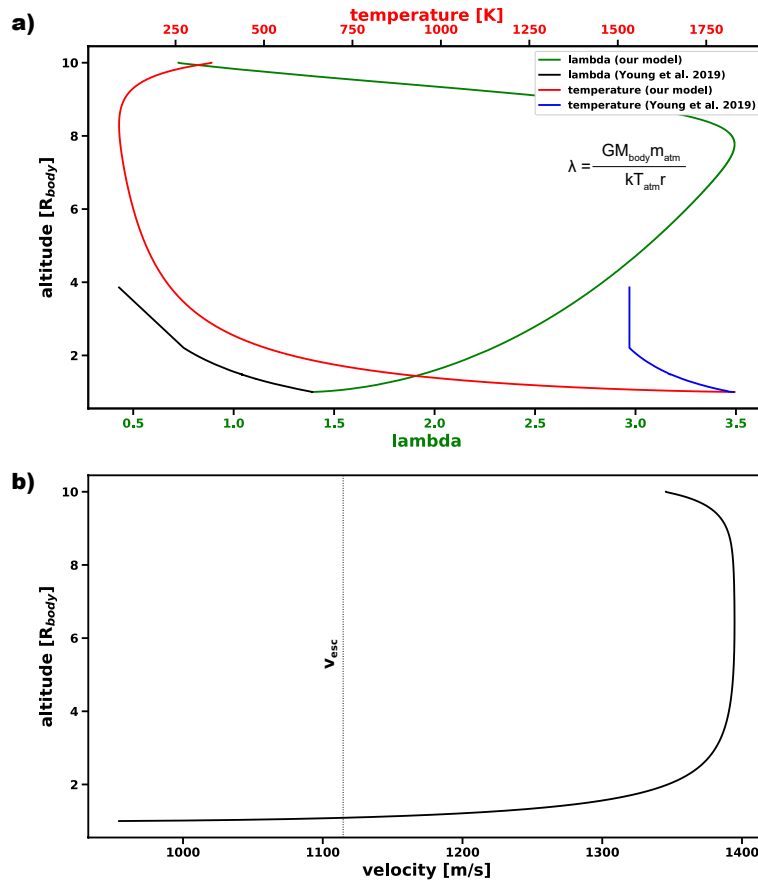


Figure 10: Another simulation run of the case described in Young et al. (2019). Here, we used the same 1D hydrodynamic atmosphere model as already described in e.g. Erkaev et al. (2015) and Erkaev et al. (2016). Panel a) shows the change of λ (green) and temperature (red) with altitude. The Jeans escape parameter λ is always below 6; close to the surface and in the thermosphere, λ is clearly below the pure hydrodynamic value of $\lambda = 2 - 3$. Clearly visible is the EUV-heating in the thermosphere as can be recognized by the clearly visible temperature inversion and accompanying decrease of λ . The blue and black lines correspond to the temperature and λ profile of the assumed hydrostatic atmosphere of Young et al. (2019). b) Bulk flow velocity of the hydrodynamically escaping silicate atmosphere.

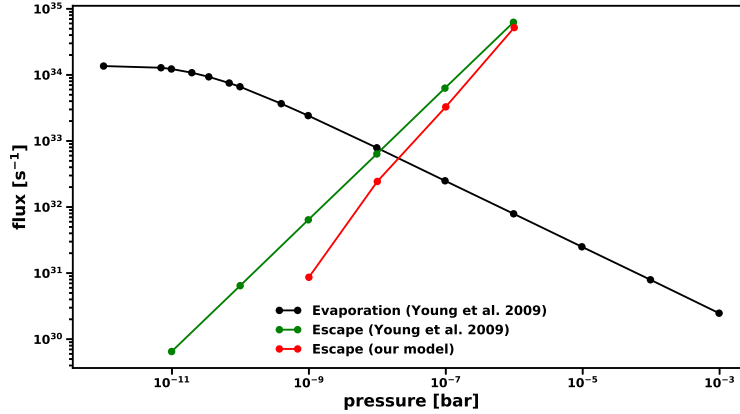


Figure 11: Outgassing from the magma ocean of the same planetary embryo (black) for different surface pressures as calculated by Young et al. (2019) and escape flux from the body as calculated by Young et al. (2019) (green) and simulated via the 1D hydrodynamic atmosphere model (red). In our case equilibrium between outgassing and escape is reached at a pressure of $\approx 2 \times 10^{-8}$ bar and a hydrodynamic escape flux of $\approx 5.5 \times 10^{32} \text{ s}^{-1}$.

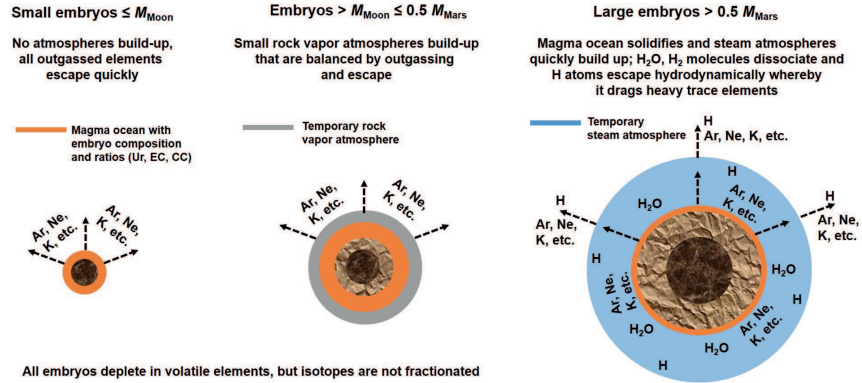


Figure 12: Illustration of the different escape regimes for small ($M_{\text{Emb}} \leq M_{\text{Moon}}$), intermediate ($M_{\text{Moon}} < M_{\text{Emb}} \leq 0.5 M_{\text{Mars}}$) and larger planetary embryos ($M_{\text{Emb}} \geq 0.5 M_{\text{Mars}}$).

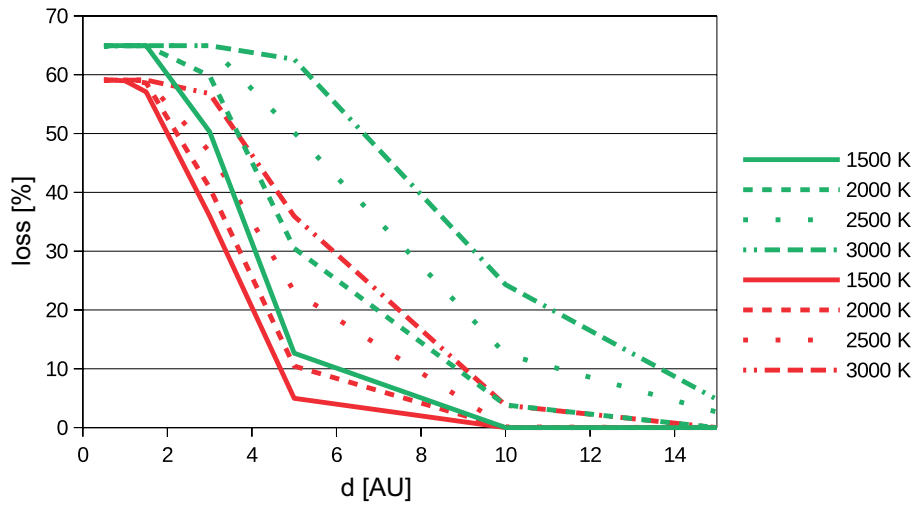


Figure 13: Relation between the orbital distance of the planetary embryo and the amount of loss for K around a slow rotator. The higher the orbital distance gets, however, the faster will a steam atmosphere condense if no shallow magma ocean remains at the surface; The values presented here are therefore maximum losses in case that the steam atmosphere doesn't condense. Green: $M_{\text{Emb}} = 1.0 M_{\text{Mars}}$ and smaller; red: $M_{\text{Emb}} = 1.5 M_{\text{Mars}}$.